## Nondestructive Monitoring of a Multilayer Pressure Vessel: Strain Gages Instrumentation

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Abstract- This article presents a method that allows verifying the safety of the operation of multilayer pressure vessels when inspection using ultrasonic testing equipment - an action commonly known as Non-Destructive Testing (NDT) - is not possible. In this context, strain gages were used to obtain strain values of the material related to stresses to which it is subjected by means of a pressurizing test. Once geometric characteristics and mechanical properties of the pressure vessel under study were known, it was possible to apply mechanics of materials concepts and obtain the biaxial state of principal stresses, thus inferring the actual thickness value of the accumulator. Moreover, it was possible to verify that although the use of strain gages takes longer than the ultrasound test, its degree of precision is remarkable and its applicability much wider, since geometric factors are not particularly concerning the shrinkage which occurs between the vessel layers. The data were also evaluated analytically and using finite element simulation, which demonstrated consistency with the results obtained by extensometry. Keywords—Extensometry; Finite Elements; Thick Wall; Pressure Vessel; Instrumentation.

#### I. INTRODUCTION

The development of industries has required the use of high pressure lines and high pressure storage accumulators, as a result creating difficulties in assembling such equipment. In this context, multilayer pressure vessels are much easier to be assembled and the generated preload stress in the process of wrapping and clamping, on the other hand, redistributes stresses and allow the pressure vessel endure higher pressure [1]. It was commonly used from the 1960's to 1990's. Consequently, many multilayer pressure vessels have been in operation for about forty years. They have been submitted to different operating environments that change structural parameters and lead to subsequent problems in operation, which are becoming major concerns [2]. The American Society of Mechanical Engineers (ASME) [3], establish the rules concerning tolerance and dimensions, however, fatigue damage from years of operation must also be considered.

When there are not many pressurizing and depressurizing cycles during its operation, for example, characterizing low-cycle operation, fatigue is not the main problem [2, 4, 5]. However, monitoring the corrosion damage in pressure vessels using nondestructive evaluation (NDE) methods is rather usual, even though the equipment operates for several years and inspect structural modification at a given frequency is important, since most failures in operations are caused by corrosion. It is well-known that there is no ideal or suitable means of corrosion detection for all corrosion mechanisms [6]

8]. In the context of pressure vessels, geometry, access (inspection nozzle) and placement are limiting factors to usual methods [9].

Short-ranging ultrasonic tests enables detection and evaluation of defects [10] based on the principles of geometrical acoustics and physical interaction of acoustic waves with the material and its defects [11, 12]. However, in a multilayer pressure vessel, contact between the outer and the inner layer deflects the wave, thus it is only possible to determine the thickness of each layer at a time through this method. The non-existence of an inspection nozzle that enables internal inspection is a hindrance to such an efficient method [9].

Strain gages allow indirect measurement in such a way that each gage consists in a fine metal grid that is stretched or shortened when the object is strained at the point to which the gage is attached [13]. By considering that the material under analysis is homogeneous, linearly elastic, isotropic and undergoes small deformations [14], it is possible to correlate the average strain of each filament submitted to the applied voltage with deformation and, consequently, with stress [15].

An NDE was performed on a hydraulic system component to apply similar pressure energy to a fluid in the path to service an open forging press for large steel parts the generated preload stress, whose capacity is 8.000 tons. In conjunction with a pump, the operation accumulator allows intermittent operation of the pump, and the presence of a control pressure vessel to homogenize the pressurized fluid, so that the accumulated energy is available to the operator, decreases pressure time. For such a purpose, it is necessary that the fluid, for example, compressed air, reaches pressure of 320 bar so that it can be characterized as a high pressure accumulator.

In this context, this work aims to perform measurements of the specific external deformations in the side and top of a pressure vessel. Calculate the principal stresses and compare them with the theoretical stresses. In addition, a computational simulation, using finite elements, will be performed to compare the obtained values.

#### II. METHODS

#### A. Assortment of instrumentation points

To carry out the NDE on strain gages, it is necessary to consider characteristics and properties of the equipment provided in Table 1. Once specific deformation is measured as

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regards the linear proceeding of steel, the only remaining step is to find a theoretical approach that best suits the study case. Multilayer pressure vessels have a typical configuration in the manufacturing process, for example, each layer is between a contact pressure as shown in [16]. As it was not considered while the project was being carried out, a safer approach suggests treating the vessel as if there was no contact pressure, made by a perfect fit. In view of this, the analysis made deal with the mechanical behavior according to considerations for a thick-walled pressure vessel [16], as shown in Figure 1.

TABLE I

MECHANICAL AND GEOMETRICAL PROPERTIES OF THE PRESSURE
VESSEL.

| Name                           | Symbol                | Value        |
|--------------------------------|-----------------------|--------------|
| Elastics Modulus               | Е                     | 200000 [MPa] |
| Poisson's Ratio                | ν                     | 0.30         |
| Inner Vessel Radius            | a                     | 477.5 [mm]   |
| Outer Vessel Radius            | b                     | 514.0 [mm]   |
| Outer Sleeve Radius            | С                     | 553,95 [mm]  |
| Inner Spherical Head<br>Radius | a1                    | 483.0 [mm]   |
| Outer Spherical Head<br>Radius | b1                    | 541.5 [mm]   |
| Operation Pressure             | OP                    | 31.38 [MPa]  |
| Hydrostatic Test               | HP                    | 47.07 [MPa]  |
| Pressure                       |                       |              |
| Fabrication Year               | 1978                  |              |
| Fabricator                     | Hitachi Co. Ltda      |              |
| Operation Temperature          | 100                   | [°C]         |
| Radiography                    | 100                   | [%]          |
| Project Code                   | Japaniese High Press. |              |
| -                              | Gas Control Law       |              |
| Head Material                  | Jis G3115 SPV-36      |              |
| Shell Material                 | Jis G3115 SPV-50      |              |

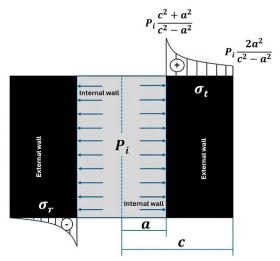


Fig. 1 Distribution of tangential and radial stresses in thick-walled tubes.

#### B. Assortment of instrumentation points

One of the most important issues in determining instrumentation points through strain gages is pressures applied on the chosen site. Nozzles and welds generate bending stresses which are undesirable given the required analytical approach [17]. For such a purpose, the points represented in Figure 2 were selected based on optimal and practical feasibility of access in order to ensure that A and B were free of bending stresses, so that there was no mutual interference whatsoever. C has two functions, since not only it allows the same analysis because the top is composed of a single layer, but its thickness is measured by ultrasound, thus allowing a comparison between a wider application of NDE and the proposed one [11, 18].

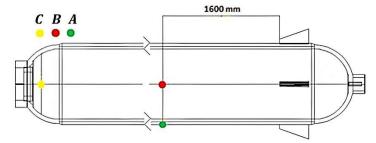


Fig. 2 Positioning of strain gauges in the pressure vessel.

#### C. Strain Gage Selection

To evaluate the evolution of deformations, the strain gage, in addition to being compatible with the analyzed material, must reflect a strain value related to a given side coordinate. As regards the triaxial state, XYZ, and terms how to calculate only the principal stress, both longitudinal and tangential, as shown in Equation (1), Equation (2) and Equation (3):

$$\sigma_{t} = \frac{p_{i}(c^{2}-a^{2})}{(c^{2}-a^{2})}$$

$$\sigma_{t} = \frac{2 p_{i}a^{2}}{(c^{2}-a^{2})}$$

$$\sigma_{l} = \frac{p a^{2}}{(c^{2}-a^{2})}$$
(3)

Once the sensor is bonded in precise alignment with the main stresses, even more considering the work in the field, it is a task that almost impossible to be achieved. Fortunately, this problem can be circumvented by using rosettes, as it can be seen in Figures 3(a) - (c). These are a combination of three unidirectional gages offset from each other at a known angle. The chosen measurement grid is the one arranged at an angle of  $45^{\circ}$  -  $120~\Omega$ .

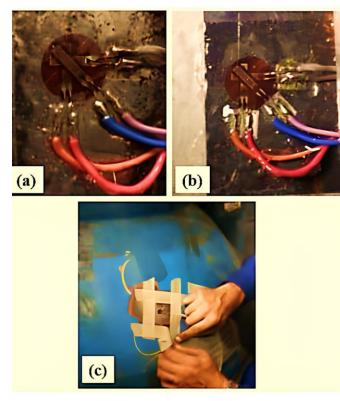


Fig. 3 Process of strain gauge bonding in pressure vessel where Figure (a) A point; Figure (b) B point and (c) C point.

Figure 2 shows the position of the rosettes on the side and top. The rosettes were positioned so that they were far from stress concentration points, so that experimental and theoretical values could be compared. The orientation of the strain gauges on each rosette in relation to the pressure vessel was: tangential (or circumferential), longitudinal and at 45° between them, as shown in the figure. To obtain the specific deformation data, the electrical strain gage technique was used, which consists of gluing electrical resistance strain gauges to a properly prepared surface. After applying the load, the strain gauge follows the deformations of the point on the surface where it is glued, varying its electrical resistance. The deformation to be measured is proportional to the variation in its electrical resistance.

Three rectangular rosettes (45°) were glued to three points (Figure 3) on the pressure vessel: two points located on the side and one point on the top. The strain gauges were connected to an electrical circuit in a ¼ Wheatstone bridge configuration, which, when connected to the HBM Quantum MX 1615 signal conditioner, provided the specific deformation values as the pressure vessel was pressurized. The electrical signal from the pressure transducer was also connected to the signal conditioner. Subsequently, deformation data were collected at the three points during the depressurization stage. The pressurization procedure was performed over a time interval of

approximately 19 hours, while the depressurization procedure was performed over a time interval of approximately 5 hours.

#### D. Data Collection

After being assembled, the pressure vessel, pressurized at 29.42 MPa (300 kgf/cm²) was depressurized, which was chosen to represent the most critical operating conditions. In this process, the signal conditioner provided the ratio between the pressure value obtained by a transducer and deformation values, three for each respective point at the following coordinates: 0 °, 45 °, 90 °. The relations represented in Equations (4) and (5) enabled the determination of the related stress based on the properties described in Table 1.

$$\sigma_{0^{\circ}} = \frac{E}{(1-\nu^{2})} (\varepsilon_{0^{\circ}} + \nu \varepsilon_{90^{\circ}})$$

$$(4)$$

$$\sigma_{90^{\circ}} = \frac{E}{(1-\nu^{2})} (\varepsilon_{90^{\circ}} + \nu \varepsilon_{0^{\circ}})$$
(5)

However, these stresses directly related to these deformations do not represent the principal stress which can be explained by the theory of thick-walled pressure vessels. As the used strain gages were rosettes and angles between the data were collected, three deformations can be related through Equations (6) and (7), thus obtaining the principal stresses.

$$\begin{split} &\sigma_{1} = E \left[ \frac{\varepsilon_{90^{\circ}} + \varepsilon_{0^{\circ}}}{2(1-\nu)} + \frac{1}{2(1+\nu)} \sqrt{(\varepsilon_{90^{\circ}} - \varepsilon_{0^{\circ}})^{2} + (2\varepsilon_{45^{\circ}} - \varepsilon_{0^{\circ}} - \varepsilon_{90^{\circ}})^{2}} \right] (6) \\ &\sigma_{2} = E \left[ \frac{\varepsilon_{90^{\circ}} + \varepsilon_{0^{\circ}}}{2(1-\nu)} - \frac{1}{2(1+\nu)} \sqrt{(\varepsilon_{90^{\circ}} - \varepsilon_{0^{\circ}})^{2} + (2\varepsilon_{45^{\circ}} - \varepsilon_{0^{\circ}} - \varepsilon_{90^{\circ}})^{2}} \right] (7) \end{split}$$

#### III. RESULTS AND DISCUSSION

A. Experimental specific deformations at points A, B and C during pressurization.

Table 2 shows the experimental results at points A, B and C according to the equipment reading. In this table, there is:

 $\epsilon_{AL}$  – Specific deformation at point A, in the longitudinal direction:

 $\epsilon A_{45}-$  Specific deformation at point A, at  $45^{\circ}$  in relation to the longitudinal direction;

 $\epsilon_{AT}$  – Specific deformation at point A, in the tangential direction.

 $\epsilon_{\text{BL}}$  – Specific deformation at point B, in the longitudinal direction;

 $EB_{45}$  – Specific deformation at point B, at  $45^{\circ}$  in relation to the longitudinal direction;

 $\epsilon_{\text{BT}}$  – Specific deformation at point B, in the tangential direction.

 $\epsilon_{\text{CL}}$  – Specific deformation at point C, in the longitudinal direction;

 $\mathcal{E}_{C45}$  – Specific deformation at point C, at 45° in relation to the longitudinal direction;

 $\epsilon_{\text{CT}}$  – Specific deformation at point C, in the tangential direction.

TABLE II
SPECIFIC DEFORMATIONS MEASURED AT POINTS A, B AND C

| Pressur          | $\mathcal{E}_{\mathrm{AL}}$ | $\mathcal{E}_{A45}$         | $\mathcal{E}_{AT}$           | $\mathcal{E}_{\mathrm{BL}}$ | $\mathcal{E}_{\mathrm{B45}}$ |
|------------------|-----------------------------|-----------------------------|------------------------------|-----------------------------|------------------------------|
| e                | (µm/m)                      | (μm/m)                      | (μm/m)                       | (μm/m)                      | (μm/m)                       |
| (MPa)            |                             |                             |                              |                             |                              |
| 20.900           | 149.38                      | 297.48                      | 444.27                       | 114.89                      | 286.17                       |
|                  | 155.00                      | 200.76                      | 1<br>455.24                  | 120.29                      | 0                            |
| 21.100           | 155.98                      | 309.76<br>0                 |                              | 120.38<br>4                 | 298.16<br>0                  |
|                  | 5<br>173.38                 | 330.93                      | 5<br>488.79                  | 136.25                      | 319.78                       |
| 22.000           | 9                           | 0                           | 8                            | 3                           | 0                            |
|                  | 185.96                      | 356.54                      | 520.04                       | 153.01                      | 342.23                       |
| 23.100           | 2                           | 0                           | 1                            | 4                           | 0                            |
| *                | 205.16                      | 383.79                      | 554.08                       | 171.72                      | 369.33                       |
| 24.100           | 4                           | 0                           | 1                            | 7                           | 0                            |
| 25 100           | 223.65                      | 412.98                      | 588.98                       | 190.59                      | 395.41                       |
| 25.100           | 1                           | 0                           | 2                            | 5                           | 0                            |
| 26.000           | 242.36                      | 439.04                      | 621.43                       | 209.60                      | 420.84                       |
| 20.000           | 8                           | 0                           | 9                            | 5                           | 0                            |
| 27.000           | 258.97                      | 463.03                      | 651.66                       | 225.31                      | 443.94                       |
| 27.000           | 8                           | 0                           | 8                            | 3                           | 0                            |
| 28.000           | 278.21                      | 492.53                      | 687.79                       | 245.41                      | 472.78                       |
| 20.000           | 7                           | 0                           | 6                            | 0                           | 0                            |
| 28.900           | 293.89                      | 515.09                      | 716.89                       | 260.06                      | 494.44                       |
|                  | 6                           | 0                           | 3                            | 5                           | 0                            |
| 29.700           | 307.80                      | 537.84                      | 744.62                       | 276.15                      | 516.47                       |
| Pressur          | 8                           | 0                           | 9                            | 2                           | 0                            |
| e                | $\mathcal{E}_{BT}$          | $\mathcal{E}_{\mathrm{CL}}$ | $\mathcal{E}_{\mathrm{C45}}$ | $\epsilon_{\rm cr}$         |                              |
| (MPa)            | $(\mu m/m)$                 | (µm/m)                      | (µm/m)                       | (µm/m)                      |                              |
|                  | 474.58                      | 160.79                      | 155.19                       | 156.73                      |                              |
| 20.900           | 1                           | 9                           | 1                            | 0                           |                              |
| 21 100           | 483.67                      | 166.67                      | 160.83                       | 162.56                      |                              |
| 21.100           | 5                           | 3                           | 1                            | 6                           |                              |
| 22,000           | 516.02                      | 199.49                      | 193.39                       | 195.51                      |                              |
| 22.000           | 7                           | 6                           | 6                            | 6                           |                              |
| 23.100           | 549.01                      | 237.30                      | 231.44                       | 233.20                      |                              |
| 25.100           | 6                           | 3                           | 7                            | 1                           |                              |
| 24.100           | 584.51                      | 271.19                      | 264.50                       | 267.10                      |                              |
| ∠¬.100           | 3                           | 2                           | 3                            | 1                           |                              |
| 25.100           | 620.31                      | 313.49                      | 306.61                       | 308.15                      |                              |
| 23.100           | 6                           | 1                           | 0                            | 7                           |                              |
| 26.000           | 653.37                      | 348.15                      | 341.17                       | 342.69                      |                              |
| 20.000           | 7                           | 3                           | 5                            | 4                           |                              |
| 27.000           | 685.42                      | 380.00                      | 372.75                       | 374.06                      |                              |
|                  | 7 727.20                    | 3<br>414.80                 | 6<br>408.34                  | 6<br>409.18                 |                              |
|                  |                             | 1 414 X()                   | 408.34                       | 409.18                      | l                            |
| 28.000           |                             |                             | _                            | 0                           |                              |
| 28.000           | 4                           | 0                           | 6                            | 9                           |                              |
| 28.000<br>28.900 | 4<br>753.28                 | 0<br>440.58                 | 432.96                       | 434.08                      |                              |
| 28.900           | 753.28<br>3                 | 0<br>440.58<br>4            | 432.96<br>4                  | 434.08<br>7                 |                              |
|                  | 4<br>753.28                 | 0<br>440.58                 | 432.96                       | 434.08                      |                              |

With the measured values (Table 2), the experimental stresses at point A and in the longitudinal and tangential directions were obtained through Equations (4) and (5). For points B and C, similar expressions were used. Table 3 shows

the results of the experimental normal stresses at points A, B and C and the subscript e, refers to experimental results, L, longitudinal and T, tangential.

### B. Experimental normal stresses at points A, B and C during pressurization

TABLE III
EXPERIMENTAL STRESSES AT POINTS A, B AND C

|                       |                       |                        |                        | 1                      |                        |                        |
|-----------------------|-----------------------|------------------------|------------------------|------------------------|------------------------|------------------------|
| Pressur<br>e<br>(MPa) | <sub>оде</sub><br>МРа | <sub>бАТе</sub><br>MPa | <sub>σВLе</sub><br>MPа | <sub>бВТе</sub><br>MPa | <sub>σCLe</sub><br>MPa | <sub>оСТе</sub><br>MPa |
| 20.900                | 62.125                | 107.49<br>2            | 56.542                 | 111.87<br>9            | 45.674                 | 45.048                 |
| 21.100                | 64.299                | 110.33<br>9            | 58.349                 | 114.24<br>0            | 47.350                 | 46.718                 |
| 22.000                | 70.336                | 118.86<br>0            | 63.969                 | 122.39<br>6            | 56.736                 | 56.124                 |
| 23.100                | 75.159                | 126.55<br>6            | 69.828                 | 130.75<br>2            | 67.530                 | 66.899                 |
| 24.100                | 81.624                | 135.30<br>3            | 76.282                 | 139.78<br>7            | 77.214                 | 76.584                 |
| 25.100                | 87.988                | 144.19<br>3            | 82.789                 | 148.90<br>0            | 89.217                 | 88.397                 |
| 26.000                | 94.242                | 152.56<br>0            | 89.147                 | 157.41<br>9            | 99.112                 | 98.273                 |
| 27.000                | 99.885                | 160.29<br>9            | 94.712                 | 165.49<br>9            | 108.18<br>1            | 107.26<br>7            |
| 28.000                | 106.49<br>6           | 169.50<br>8            | 101.88<br>4            | 176.00<br>6            | 118.14<br>4            | 117.28<br>1            |
| 28.900                | 111.86<br>0           | 176.93<br>7            | 106.82<br>4            | 182.70<br>4            | 125.45<br>3            | 124.45<br>3            |
| 29.700                | 116.74<br>7           | 183.95<br>0            | 112.28<br>7            | 190.18<br>8            | 135.29<br>8            | 134.23<br>4            |

C. Experimental principal stresses at points A, B and C during pressurization.

The principal stresses at points A, B and C were calculated using the equations obtained from the analysis of the rectangular rosette, Equation (6) and (7). They are shown in Table 4.

| Pressur<br>e<br>(MPa) | <sub>бА1е</sub><br>MPa | <sub>бА2е</sub><br>MPa | <sub>овіе</sub><br>MPa | <sub>бВ2е</sub><br>MPa | <sub>σС1е</sub><br>MPа | <sub>σС2е</sub><br>MPа |
|-----------------------|------------------------|------------------------|------------------------|------------------------|------------------------|------------------------|
| 20.900                | 107.49<br>2            | 62.125                 | 111.91<br>0            | 56.511                 | 45.994                 | 44.729                 |
| 21.100                | 110.34<br>7            | 64.290                 | 114.24<br>6            | 58.342                 | 47.697                 | 46.371                 |
| 22.000                | 118.86<br>0            | 70.336                 | 122.41                 | 63.953                 | 57.133                 | 55.728                 |
| 23.100                | 126.56<br>2            | 75.153                 | 130.78<br>2            | 69.798                 | 67.880                 | 66.550                 |
| 24.100                | 135.31<br>1            | 81.616                 | 139.81<br>6            | 76.253                 | 77.680                 | 76.118                 |
| 25.100                | 144.21<br>2            | 87.969                 | 148.93<br>6            | 82.753                 | 89.574                 | 88.040                 |

| 26.000 | 152.58<br>1 | 94.221 | 157.45<br>9 | 89.108 | 99.469      | 97.916      |
|--------|-------------|--------|-------------|--------|-------------|-------------|
| 27.000 | 160.32<br>2 | 99.862 | 165.54<br>3 | 94.669 | 108.52<br>5 | 106.92<br>3 |
| 28.000 | 169.54      | 106.46 | 176.06      | 101.82 | 118.42      | 117.00      |
|        | 2           | 2      | 4           | 5      | 1           | 5           |
| 28.900 | 176.97      | 111.82 | 182.75      | 106.77 | 125.79      | 124.11      |
|        | 1           | 6      | 1           | 7      | 1           | 5           |
| 29.700 | 183.99      | 116.69 | 190.23      | 112.23 | 135.65      | 133.87      |
|        | 7           | 9      | 8           | 6      | 6           | 6           |

It can be observed that the values found for the principal stresses are very close to the values of the longitudinal and tangential stresses for points A, B and C, proving that the longitudinal and tangential classifications are already the main ones.

# D. Experimental specific deformations at points A, B and C during depressurization and the respective tensions of interest

Now, the main stresses for depressurization are presented, Table 5. The method used is identical to that used in pressurization, with the results shown above, in Table 2.

TABLE V SPECIFIC DEFORMATIONS MEASURED AT POINTS A, B AND C - DEPRESSURIZATION

| Pressure<br>* | E <sub>AL</sub> *       | ε <sub>A45*</sub> | $\mathcal{E}_{AT^*}$ | ε <sub>вι.*</sub>       | Ε <sub>В45*</sub> |
|---------------|-------------------------|-------------------|----------------------|-------------------------|-------------------|
| (MPa)         | (μm/m)                  | (µm/m)            | (µm/m)               | (μm/m)                  | (μm/m)            |
| 0.400         | 2.920                   | 4.620             | 7.250                | 3.250                   | 4.880             |
| 1.520         | 14.440                  | 20.560            | 29.100               | 15.540                  | 23.430            |
| 3.900         | 48.060                  | 75.910            | 104.65<br>0          | 46.460                  | 76.280            |
| 6.730         | 82.780                  | 131.23<br>0       | 180.06<br>0          | 74.020                  | 126.37<br>0       |
| 10.300        | 126.32<br>0             | 201.46<br>0       | 275.62<br>0          | 111.67<br>0             | 193.14<br>0       |
| 14.250        | 176.06<br>0             | 282.61<br>0       | 383.46<br>0          | 154.72<br>0             | 265.28<br>0       |
| 17.250        | 211.21<br>0             | 339.71<br>0       | 462.69<br>0          | 184.54<br>0             | 319.07<br>0       |
| 19.100        | 232.66                  | 376.02<br>0       | 513.32<br>0          | 203.64                  | 353.12<br>0       |
| 20.400        | 249.00                  | 402.88<br>0       | 550.50<br>0          | 215.32                  | 376.07<br>0       |
| 21.900        | 260.70                  | 427.21<br>0       | 584.81<br>0          | 226.34                  | 397.27<br>0       |
| 22.900        | 267.53<br>0             | 442.00<br>0       | 606.64               | 231.77                  | 410.81            |
| 28.500        | 325.27<br>0             | 544.32<br>0       | 747.71<br>0          | 278.77<br>0             | 497.87<br>0       |
| Pressure<br>* | ε <sub>BT*</sub> (μm/m) | E <sub>CL*</sub>  | £ <sub>C45*</sub>    | ε <sub>CT*</sub> (μm/m) |                   |
| (MPa)         | (μπ/π)                  | (µm/m)            | (µm/m)               | (μιι/ΙΙΙ)               |                   |
| 0.400         | 7.670                   | 2.610             | 2.480                | 2.540                   |                   |
| 1.520         | 32.860                  | 12.160            | 11.650               | 12.020                  |                   |
| 3.900         | 112.68<br>0             | 53.890            | 51.270               | 49.210                  |                   |
| 6.730         | 188.31                  | 90.690            | 85.130               | 80.630                  |                   |

| 10.300 | 285.98 | 138.67 | 130.67 | 123.18 |
|--------|--------|--------|--------|--------|
| 10.300 | 0      | 0      | 0      | 0      |
| 14.250 | 395.23 | 201.99 | 191.74 | 180.97 |
| 14.230 | 0      | 0      | 0      | 0      |
| 17.250 | 474.03 | 239.92 | 228.53 | 215.75 |
| 17.230 | 0      | 0      | 0      | 0      |
| 19.100 | 525.89 | 266.75 | 255.08 | 241.62 |
| 19.100 | 0      | 0      | 0      | 0      |
| 20.400 | 560.94 | 286.83 | 275.16 | 262.27 |
| 20.400 | 0      | 0      | 0      | 0      |
| 21.900 | 596.34 | 302.57 | 292.13 | 279.88 |
| 21.900 | 0      | 0      | 0      | 0      |
| 22.900 | 618.09 | 319.85 | 309.63 | 297.74 |
| 22.900 | 0      | 0      | 0      | 0      |
| 28.500 | 758.71 | 404.63 | 397.19 | 387.49 |
| 28.500 | 0      | 0      | 0      | 0      |

Once the deformations at points A, B and C have been obtained, the normal stresses and principal stresses for depressurization can be calculated in the same way as was done for pressurization. As shown in Tables 6 and 7, respectively.

 $TABLE\ VI$  Experimental stresses at points A, B and C - despressurization

| Pressure | σALe*   | σATe*   | σBLe*   | σBTe*   | σCLe*   | σCTe*   |
|----------|---------|---------|---------|---------|---------|---------|
| (MPa)    | MPa     | MPa     | MPa     | MPa     | MPa     | MPa     |
| 0.400    | 1.120   | 1.786   | 1.220   | 1.900   | 0.741   | 0.730   |
| 1.520    | 5.092   | 7.348   | 5.582   | 8.247   | 3.465   | 3.444   |
| 3.900    | 17.463  | 26.169  | 17.640  | 27.828  | 15.089  | 14.369  |
| 6.730    | 30.065  | 45.032  | 28.684  | 46.267  | 25.248  | 23.700  |
| 10.300   | 45.935  | 68.905  | 43.399  | 70.216  | 38.599  | 36.216  |
| 14.250   | 63.978  | 95.885  | 60.064  | 97.065  | 56.325  | 53.092  |
| 17.250   | 76.927  | 115.616 | 71.813  | 116.350 | 66.955  | 63.236  |
| 19.100   | 84.979  | 128.158 | 79.430  | 129.007 | 74.557  | 70.691  |
| 20.400   | 91.022  | 137.407 | 84.308  | 137.480 | 80.332  | 76.554  |
| 21.900   | 95.856  | 145.719 | 89.064  | 145.987 | 84.953  | 81.462  |
| 22.900   | 98.796  | 150.967 | 91.692  | 151.125 | 89.928  | 86.526  |
| 28.500   | 120.787 | 185.778 | 111.293 | 185.130 | 114.478 | 111.842 |

TABLE VII
EXPERIMENTAL PRINCIPAL STRESSES AT POINTS A, B AND CDESPRESSURIZATION

| Pressure<br>*<br>(MPa) | <sub>од1е*</sub><br>МРа | <sub>бА2е*</sub><br>MPa | <sub>ов1е*</sub><br>МРа | <sub>σВ2е*</sub><br>MPа | σCle*<br>MPa | σC2e*<br>MPa |
|------------------------|-------------------------|-------------------------|-------------------------|-------------------------|--------------|--------------|
| 0.400                  | 1.794                   | 1.112                   | 1.912                   | 1.208                   | 0.751        | 0.720        |
| 1.520                  | 7.363                   | 5.077                   | 8.252                   | 5.577                   | 3.523        | 3.386        |
| 3.900                  | 26.169                  | 17.462                  | 27.853                  | 17.615                  | 15.091       | 14.366       |
| 6.730                  | 45.032                  | 30.065                  | 46.298                  | 28.653                  | 25.252       | 23.696       |
| 10.300                 | 68.905                  | 45.935                  | 70.244                  | 43.370                  | 38.599       | 36.215       |
| 14.250                 | 95.891                  | 63.972                  | 97.125                  | 60.003                  | 56.326       | 53.091       |
| 17.250                 | 115.621                 | 76.922                  | 116.405                 | 71.758                  | 66.958       | 63.233       |
| 19.100                 | 128.163                 | 84.974                  | 129.072                 | 79.365                  | 74.562       | 70.686       |
| 20.400                 | 137.412                 | 91.017                  | 137.545                 | 84.243                  | 80.334       | 76.551       |
| 21.900                 | 145.728                 | 95.846                  | 146.069                 | 88.982                  | 84.958       | 81.456       |
| 22.900                 | 150.978                 | 98.785                  | 151.205                 | 91.612                  | 89.933       | 86.522       |
| 28.500                 | 185.801                 | 120.765                 | 185.269                 | 111.154                 | 114.490      | 111.830      |

#### E. Theoretical stresses at points A, B and C

To obtain the theoretical stresses, the thick-walled tube theory was used, since the ratio between the tube thickness and the internal radius is greater than 10%. In this case, the tangential and radial stresses behave as shown in Figure 1. On

the internal surface there is a triple stress state (longitudinal, radial and tangential stress), while on the external surface there is a biaxial state (longitudinal and tangential stress).

The tangential and radial stresses in the side are given by:

- On the internal surface, the tangential stress is equivalent to Equation (1) and the radial stress has a value equal to pi, where pi is the internal pressure;
- On the outer surface, the tangential stress is equivalent to Equation (2) and the radial stress has a value equal to zero;
- Longitudinal stresses are obtained considering uniform stress distribution, according to Equation (3).
- In the spherical top, the tangential stresses were obtained by the thin-wall theory, Equation (8):

$$\sigma_t = \frac{p_l a_1}{2t} \tag{8}$$

Table 8 shows the tangential and longitudinal stresses calculated according to the theoretical expressions above.

TABLE VIII
THEORICAL PRINCIPAL STRESSES AT POINTS A. B AND C

| Pressur |                        |                        |                       |                        | σCLt   | σCTt   |
|---------|------------------------|------------------------|-----------------------|------------------------|--------|--------|
| e       | <sub>одLt</sub><br>MPa | <sub>бАТt</sub><br>MPa | <sub>овь</sub><br>МРа | <sub>σВТt</sub><br>MРа | MPa    | MPa    |
| (MPa)   | IVII a                 | IVII a                 | IVII a                | IVII a                 |        |        |
| 0.000   | 0.000                  | 0.000                  | 0.000                 | 0.000                  | 0.000  | 0.000  |
| 2.000   | 5.783                  | 11.566                 | 5.783                 | 11.566                 | 8.254  | 8.254  |
| 4.000   | 11.56<br>6             | 23.132                 | 11.56<br>6            | 23.132                 | 16.507 | 16.507 |
| 6.000   | 17.34<br>9             | 34.698                 | 17.34<br>9            | 34.698                 | 24.761 | 24.761 |
| 8.000   | 23.13                  | 46.264                 | 23.13<br>2            | 46.264                 | 33.014 | 33.014 |
| 10.000  | 28.91<br>5             | 57.830                 | 28.91<br>5            | 57.830                 | 41.268 | 41.268 |
| 12.000  | 34.98                  | 69.396                 | 34.69<br>8            | 69.396                 | 49.522 | 49.522 |
| 14.000  | 40.48<br>1             | 80.962                 | 40.48<br>1            | 80.962                 | 57.775 | 57.775 |
| 16.000  | 46.26<br>4             | 92.528                 | 46.26<br>4            | 92.528                 | 66.029 | 66.029 |
| 18.000  | 52.04<br>7             | 104.09<br>3            | 52.04<br>7            | 104.09                 | 74.282 | 74.282 |
| 20.000  | 57.83<br>0             | 115.65<br>9            | 57.83<br>0            | 115.65<br>9            | 82.536 | 82.536 |
| 22.000  | 63.61                  | 127.22<br>5            | 63.61<br>3            | 127.22<br>5            | 90.789 | 90.789 |
| 24.000  | 69.39<br>6             | 138.79<br>1            | 69.39<br>6            | 138.79<br>1            | 99.043 | 99.043 |
| 26,000  | 75.17                  | 150.35                 | 75.17                 | 150.35                 | 107.29 | 107.29 |
| 20.000  | 9                      | 7                      | 9                     | 7                      | 7      | 7      |
| 28.000  | 80.96                  | 161.92                 | 80.96                 | 161.92                 | 115.55 | 115.55 |
| 28.000  | 2                      | 3                      | 2                     | 3                      | 0      | 0      |
| 20,000  | 86.74                  | 173.48                 | 86.74                 | 173.48                 | 123.80 | 123.80 |
| 30.000  | 5                      | 9                      | 5                     | 9                      | 4      | 4      |

Once the main stress was determined according to the data obtained by the strain gages analyses and Equations (6) and (7), similar values at points A and B were verified, thus the instrumentation was validated as successful. At Point C, values were identical, which is typical of spherical geometry. In Figure 4, the experimental values can be compared with those obtained from Equations (2) and (3) as for the pressure variation in the test.

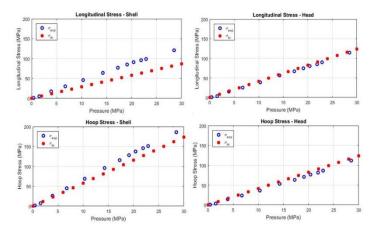


Fig. 4 Tangential and longitudinal stresses at points A and C. (At point B, stresses similar to those at point A act) – despressurization

The experimental linearity and similarity between the two analyzes allow inferring an indirect relation for safety inspection of the pressure vessel. To estimate the side thickness, Equation (2) was used, which equates the values of experimental tangential stress obtained from depressurization test. Therefore, the value to be obtained from this equation will be the internal radius by considering that the external radius, in addition to being measured, must be subjected throughout operating time so that a very mild corrosion effect is produced. From this relation, once the internal radius (a) value is known, the external radius (c) is obtained.

When all pressures concerning the geometry of the pressure accumulator were analyzed, a minimum value of 33.9 mm of internal thickness was reached. With due consideration of design as regards safe working pressure conditions, experimental tangential stresses on the side and top (on the outer surface) corresponding to an internal pressure of 29.7 MPa are already close to the material stress value established by the JIS standard (SPV 355, with  $\sigma_{adm}=160$  MPa and SPV490, with  $\sigma_{adm}=195$  MPa). Furthermore, on the inner surface, tangential stresses are greater than on the outer surface (if there is no contact pressure between the vessels). Thus, according to experimental stresses, it is considered that pressure of 29.7 MPa is already on the recommended limit. The determined values can be compared with properties of pressure vessel materials show.

To complement the evaluation, a simulation was performed using the finite element tool, Figure 5, through the Inventor software. Finite elements are a powerful tool in engineering,

used in various fields, bringing together theory and practice through computational simulation [19 - 21]. With this, the principal stresses were obtained at the points studied and the coherence in terms of magnitude of the experimentally and theoretically obtained values was observed. This ratifies the method using extensometry to measure stresses in pressure vessels

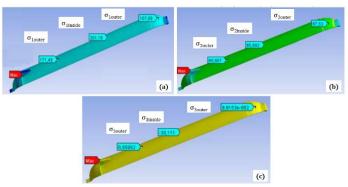


Fig. 5 Principal stresses determined by means of finite elements

According to [22], NDT techniques are essential for quality assessment and performance prediction. They are also used to locate damage and identify failure modes of pressure vessels. In this work, the use of NDT, specifically the application of extensometry, was essential to obtain the deformations and stresses generated in the pressure vessel when a load is applied. Furthermore, the use of FEM is common in the assessment of the structural integrity of pipelines and pressure vessels, due to its ability to model defects and predict stress loads. It is a tool that, together with other NDT techniques, enables the assessment of structures and monitoring in operation [22 – 25].

#### IV. CONCLUSIONS

From the analysis of the data obtained, it was observed that the measurements performed by means of extensometry presented results very close to those predicted by the theoretical equations, evidencing the reliability of this technique for monitoring stresses in multilayer pressure vessels. Furthermore, the application of the finite element method proved to be an effective and accurate tool in obtaining stress fields, reinforcing its relevance in structural analyses and in the development of engineering projects. In this way, the pressure vessel can be monitored and its safety in operation ensured. It is worth mentioning that for this case it was necessary to use the multilayer theory, a difficult case to be addressed in operation, demonstrating the applicability between theory and practice.

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#### REFERENCES

- [1] Z. Liang, J. Nan, and L. Sijia, "Experimental study of integrated multilayer clamping high pressure vessel," *J. Press. Vessel Technol.*, vol. 133, no 6, pp. 61206 61212, December 2011.
- [2] M. Zhao and R. Parkinson, "Risk-based operations assessment of a multilayered vessel under cyclic loading conditions," *J. Press. Vessel Technol.*, vol. 136, no 2, pp. 489-496, August 2013.
- [3] ASME, "Boiler and Pressure Vessel Code", Section VIII, Division 1: Rules for Construction of Pressure Vessels, American Society of Mechanical Engineers, 2015.
- [4] H. C. Sampaio, R. A. Dias and J. A. P. Balestieri, "Sustainable Urban Energy Planning: The Case Study of a Tropical City," *Appl. Energy*, vol. 104, pp. 924–935, April 2013.
- [5] P. V. Piping and S. Eng, "Fatigue Abstracts," *Situ*, vol. 63, no 10, pp. 2–3, November 1995.
- [6] A. R. Faria Neto, A. P. N. Souza, R. R. Passos, M. S. Pereira, L. A. Pocrifka, R. F. Brambilla Souza and E. Costa Rios, "Evaluation of Paullinia Cupana as a green corrosion inhibitor for carbon steel utilizing gravimetric and electrochemical noise techniques", Materials Research Express, vol. 6, no. 7, pp. 076522, April 2019.
- [7] V. S. Agarwala, P. L Reed, S. Ahmad, "Corrosion detection and monitoring – A review", *Nace Corrosion* p. NACE-00271, 2000.
- [8] E. Bardal and J. M. Drugli, "Corrosion detection and diagnosis", Materials science and engineering, vol. 3, pp. 144-164, 2004.
- [9] M. G. Lozev, R. W. Smith and B. B. Grimmett, "Evaluation of Methods for Detecting and Monitoring of Corrosion Damage in Risers," *J. Press. Vessel Technol.*, vol. 127, no 3, pp. 244-254, August 2005.
- [10] J. Prager, J. Kitze, C. Acheroy, D. Brackrock, G. Brekow and M. Kreutzbruck, "SAFT and TOFD A Comparative Study of Two Defect Sizing Techniques on a Reactor Pressure Vessel Mock-Up," J. Nondestruct. Eval., vol. 32, no 1, pp. 1–13. August 2013.
- [11] Neiss, G., "Nondestructive inspection of nozzle inner corner at boiling water reactor pressure vessels with ultrasound-flaw detection and crack depth determination," *Nucl. Engeneering Des.*, vol. 87, pp. 175–184, July 1985.
- [12] K. H. Matlack, J. Y. Kim, J. J. Wall, J. Qu, L. J. Jacobs and M. A. Sokolov, "Sensitivity of Ultrasonic Nonlinearity to Irradiated, Annealed, and Re-Irradiated Microstructure Changes in RPV Steels," J. Nucl. Mater., vol. 448, no 1–3, pp. 26–32, May 2014.
- [13] N. T. Younis and B. Kang, "Averaging Effects of a Strain Gage," J. Mech. Sci. Technol., vol. 25, no. 1, pp. 163–169. March 2011.
- [14] H. Sarangi, K. S. R. K. Murthy and D. Chakraborty, "Optimum Strain Gage Locations for Accurate Determination of the Mixed Mode Stress Intensity Factors," *Eng. Fract. Mech.*, vol. 88, pp. 63–78, July 2012
- [15] C. Carral, N. Charvin, H. Trouvé and P. Mélé, "An Experimental Analysis of PEMFC Stack Assembly Using Strain Gage Sensors," *Int. J. Hydrogen Energy*, vol. 39, no 9, pp. 4493–4501, March 2014.
- [16] V. Féodossiev, Tubos de paredes gruesas y discos. In: P. G. Mora, Résistance Des Matériaux, Éditions de la Paix, pp. 301 – 305, 1980.
- [17] S. Marlette, P. Freyer, M. Smith, A. Goodfellow, X. Pitoiset, B. Voigt, R. Rishel, and E. Kingston, "Simulation and Measurement of Through–Wall Residual Stresses in a Structural Weld Overlaid Pressurizer Nozzle," J. Press. Vessel Technol., vol. 136, no 5, pp. 1425- 1433, January 2011.
- [18] B. H. Chance and D. E. Bray, "Nondestructive Monitoring of Stress Relaxation in Welded Steel Plates," *J. Press. Vessel Technol.*, vol. 124, no 3, pp. 343-348, July 2002.
- [19] F. C. S. De Almeida, G. C. R. Oliveira and E. S. Guidi, "Study of the Stress Concentration Factor in PPS/C Type Composite Using FEA and Analytical Method," SAE Technical Paper, no 2024-36-0072, 2024.
- [20] C. S. S. Rigo, A. D. R. D. F. Neto, F. J. Grandinetti, T. S. Castro, E. X. Dias, and M. S. Martins, "Development of the Steering System for a Formula SAE Prototype," SAE Technical Paper, no 2024-36-0099, 2024.
- [21] L. O. Gomes, F. J. Grandinetti, M. S. Martins, A. M. Soares, A. R. de Faria Neto, T. S. Castro and L. F. Almeida, "Brake Pedal Sizing and

- Preliminary Design of Balance Bar in the Brake of a SAE Formula Type Vehicle," *SAE Technical Paper*, no 2024-36-0054, 2024.
- [22] W. Zhou, J. Wang, Z. Pan, J. Liu, L. Ma, J. Zhou, Y. Su, "Review on optimization design, failure analysis and non-destructive testing of composite hydrogen storage vessel", *International journal of hydrogen energy*, vol. 47, no. 91, pp. 38862-38883, november 2022.
- [23] S. D. Vijaya Kumar, M. Lo Yin Kai, T. Arumugam, S. Karuppanan, "A review of finite element analysis and artificial neural networks as failure pressure prediction tools for corroded pipelines", *Materials*, vol. 14, no. 20, pp. 6135, october 2021.
- [24] G. C. R. d. Oliveira, V. A. Rosario de Oliveira, C. A. Alvarado Silva, E. S. Guidi and F. d. A. Silva, "Modeling the Stress Field in MSLA-Fabricated Photosensitive Resin Components: A Combined Experimental and Numerical Approach." *Modelling*, vol 6, no 3, january 2025.
- [25] C. A. Alvarado Silva, , A. C. De Souza, G. C. R. de Oliveira, F. Azevedo Silva, B. U. Rubio, "Experimental Analysis of the Failure of the Bearing Support of a Cardan Shaft | Análise Experimental da Falha do Suporte de Mancal de um Eixo Cardan", Proceedings of the LACCEI international Multi-conference for Engineering, Education and Technology, july 2022.